Partial Banana Mapping: impact probability estimation and precovery of asteroids

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1. Impact probability estimation



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Asteroid's position uncertainty



Uncertainty along the orbit



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 $V_c\pm\Delta V$



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The Monte Carlo method

From the cloud of Virtual Asteroids (VAs) at the epoch of observations we randomly choose VAs according to the distribution function. The impact probability P_{MC} is:

$$P_{MC} = \frac{k}{n}$$

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k is a number of VAs, which collide with the Earth, n is a number of all considered VAs.

The mean square error δ_{MC} :

$$\delta_{MC} = \frac{\sqrt{P_{MC}(1 - P_{MC})}}{\sqrt{n}} \approx \frac{1}{\sqrt{k}} * 100\%$$

Disadvantages:

Time consuming.

Impractical for searching for possible impactors.

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- A linear relation between errors of orbital parameters at different epochs is assumed
- Consequently errors of orbital parameters have a Gaussian distribution at all considered times
- Uncertainty region is constructed from a covariance matrix at each time
- The covariance matrix is computed by propagation of the nominal orbit with variational equations
- Extremely important the choice of orbital parameters (coordinates and velocities / Keplerian elements ...)

Advantages: Fast

Disadvantages: Gravitational perturbations from major planets



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Target plane method

Integrate the asteroid's orbit until it enters the Earth's sphere of action (\approx 929,000 km) Target plane — plane perpendicular to incoming asymptote of the hyperbola (\vec{v}_{∞})



Target plane method



1. Integrate the asteroid's

orbit until it enters the Earth's sphere of influence.

- 2. The uncertainty region is represented as an ellipsoid
- 3. Map the uncertainty ellipsoid onto the target plane
- 4. Impact probability

$$P = \frac{1}{2\pi |\mathrm{det}\mathbf{L}|^{\frac{1}{2}}} \int_{S_{R'_{\oplus}}} e^{-(\boldsymbol{u}^{\mathrm{T}\mathbf{L}^{-1}}\boldsymbol{u})/2} \mathrm{d}\boldsymbol{u}$$

where $S_{R'_{\oplus}}$ is the projection of the Earth (circle) with radius $R'_{\oplus} = R_{\oplus} \sqrt{1 + \frac{v_s^2}{v_{\infty}^2}}$. v_s is the escape velocity ($\approx 11.186 \text{ km/s}$)



Problems

- The uncertainty region is mostly stretched along the nominal asteroid's orbit
- The region is very thin (pprox several times smaller than the radius of the Earth)
- But it is very long (can be the whole orbit and even more)
- If the 3- σ uncertainty is the whole orbit the impact probability value can be up to $\approx 10^{-4}$



Integrate asteroid's orbit until the Earth is close to the osculating orbit.

Construct a banana-shaped uncertainty region.

Find closest to the Earth point B along main line of the banana and compute probability for B by target plane method.

Compute how to change the original orbit so that asteroid goes to point B and collide.



 A — nominal position of an asteroid at a possible collision

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— probable position of an asteroid which is closest to the $\mathsf{Earth}_{\mathsf{c}}$

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A — nominal position of an asteroid at a possible collision

B — probable position of an asteroid which is closest to the Earth $_{\scriptscriptstyle C}$



 x_0 — state vector at epoch of observations x — state vector at time of possible collision E — Keplerian elements at time of possible collision E^{min} — Keplerian elements of point B

$$\mathbf{Q} = \left[\frac{\partial E}{\partial x}\right] \left[\frac{\partial x}{\partial x_0}\right]$$

$$Q^{-1}[E^{min} - E] = [x_0^{min} - x_0]$$

Use x_0^{min} , integrate this orbit and compute impact probability Multiply probability by a factor $\exp(-\frac{1}{2}[x_0^{min}-x_0]^T C^{-1}[x_0^{min}-x_0])$



Comparison of impact probabilities

Designation	signation $P_{MC} \pm 3\sigma_{MC}$		P_{PBM}	P_{IPBM}
2006 JY26	$(5.6 \pm 1.7) \cdot 10^{-5}$	$1.0\cdot 10^{-4}$	$1.0\cdot 10^{-4}$	$6.0 \cdot 10^{-5}$
2006 QV89	$(1.8 \pm 0.1) \cdot 10^{-3}$	$2.0\cdot 10^{-3}$	$1.8\cdot10^{-3}$	$1.8\cdot10^{-3}$
2010 UK	$(3.1 \pm 0.7) \cdot 10^{-3}$	$1.8\cdot 10^{-3}$	$1.8\cdot 10^{-3}$	$2.9\cdot 10^{-3}$
2011 AG5	$(5.3 \pm 1.3) \cdot 10^{-4}$	$4.2 \cdot 10^{-4}$	$5.7\cdot10^{-4}$	$5.5\cdot10^{-4}$
2007 VK184	$(6.2 \pm 2.0) \cdot 10^{-6}$	$2.7\cdot 10^{-5}$	$2.6\cdot\mathbf{10^{-5}}$	$8.8\cdot 10^{-6}$
2007 VE191	$(6.4 \pm 1.0) \cdot 10^{-4}$	0.0	$6.8\cdot10^{-4}$	$7.0\cdot10^{-4}$
2008 JL3	$(3.0 \pm 0.4) \cdot 10^{-4}$	$7.5\cdot 10^{-4}$	$4.0\cdot 10^{-4}$	$3.3\cdot10^{-4}$
2014 WA	$(3.5 \pm 2.4) \cdot 10^{-7}$	0.0	$5.4 \cdot 10^{-7}$	$5.3\cdot10^{-7}$
2009 JF1	$(7.4 \pm 1.2) \cdot 10^{-4}$	$7.3\cdot10^{-4}$	$8.0\cdot10^{-4}$	$7.6\cdot10^{-4}$
2012 MF7	$(3.1 \pm 0.8) \cdot 10^{-4}$	0.0	$4.8\cdot10^{-4}$	$3.2\cdot10^{-4}$
2008 CK70	$(6.4 \pm 1.0) \cdot 10^{-4}$	$5.8\cdot 10^{-4}$	$6.9\cdot10^{-4}$	$6.5\cdot10^{-4}$
2005 BS1	$(1.4 \pm 0.2) \cdot 10^{-4}$	0.0	$1.4 \cdot 10^{-4}$	$1.4 \cdot 10^{-4}$
2005 QK76	$(4.3 \pm 0.9) \cdot 10^{-5}$	0.0	$4.1 \cdot 10^{-5}$	$4.3 \cdot 10^{-5}$
2007 KO4	$(7.3 \pm 4.0) \cdot 10^{-7}$	0.0	$2.2\cdot\mathbf{10^{-6}}$	$9.7 \cdot 10^{-7}$
Apophis 2036	$(1.4 \pm 0.8) \cdot 10^{-5}$	0.0	0.0	$1.2 \cdot 10^{-5}$
2010 RF12	$(0.0 \pm 1.7) \cdot 10^{-6}$	$5.1\cdot10^{-2}$	$4.9\cdot10^{-2}$	0.0

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Even if we are mistaken by 50 days in the time of a possible collision the improved approach computes the probabilities accurately! (need 5 iteration)



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2. Finding known NEOs on old photographic plates (precovery)



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NAROO machine



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- New Astrometric Reduction of Old Observations (NAROO) program
- About 22000 photographic plates
- From 1935 until 2002
- From: Observatoire de la Côte d'Azur Nice (OCA Nice), OCA Caussols, Palomar Mountain, European Southern Observatory La Silla, Laboratoire d'Astrophysique de Bordeaux.
- Maximal visual magnitude $\lesssim 20^m$
- Field of View of the plates $\approx 6^\circ$
- The scanned image is about 11 Gb!

Can be known asteroids there!



- If the orbit of an asteroid is not well precise it is difficult to associate it with old observations.
- Because of the uncertainty the nominal asteroid's position on the sky can be far from the real one (not on the plate)
- Modify Partial Banana Mapping to this problem.
- Instead of projection onto the target plane here we will project onto the celestial sphere.
- Take 51 virtual asteroids on the main line of curvilinear uncertainty region.
- Map each of them with uncertainty ellipsoids and direction of movement



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Results for 2015 UM67 precovery on 3 March 1990.

The orange dot is the nominal position of the object.

Blue dots and lines — possible positions of the object at that time.

Red line — actual observation.

Green line — uncertainty ellipse computed from covariance matrix in RA and DEC



Precovery for 2013 NJ10

Blue dots and lines — possible positions of the object at that time. Red line — actual observation.



- Compute the probability that an asteroid can be found on the plate.
- Compute the length of the uncertainty and the visual magnitude.
- Choose the ones with higher probability, higher uncertainty and lower magnitude.

Asteroid	Date	Probability	Magnitude	Length, arcsec
1999GL4	1995-Oct-19 08:33:00	83.7	18.6	77939.3
2014PL51	1954-Sep-06 10:19:12	55.1	17.4	111613.6
2018VQ3	1988-Jun-13 04:48:00	97.8	18.9	79767.9
2016HP19	1950-04-13 08:31:32	100.0	19.0	12137.2
2019BJ5	1954-Jun-28 07:21:58	45.9	16.7	33981.0
*2010EX11	1953-Mar-14 10:35:00	86.4	19.0	2556.0

*We hope what we found is this asteroid.

Precovery for 2014PL51



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- Partial banana mapping is better than other linear methods for impact probability computation.
- Partial banana mapping method is suitable for precovery of asteroids.

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Thank you for attention!



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There are monitoring systems that calculate impact probabilities of asteroids with the Earth: CNEOS, NASA Jet Propulsion Laboratory and NEOCC, European Space Agency. On April 11, 2024:

- ESA has 1601 Near-Earth Objects
- NASA has 1716 Near-Earth Objects
- Sometimes both systems have different number of possible collisions and different impact probability values for the same asteroid.



NEOForCE: scheme

Apply Partial Banana Mapping to each virtual asteroid on the Line of Variations.



Results. Asteroid 2008 JL3

Date	NEOForCE	NASA	ESA	Date	NEOForCE	NASA	ESA
2027-05-01	3.3e-4	1.5e-4	1.49E-4	2067-04-30	1.1e-8	-	-
2029-04-30	3.6e-8	-	-	2071-04-30	1.1e-6	2.6e-6	3.04E-6
2030-04-30	-	3.0e-8	1.59E-8	2075-05-02	1.9e-7	1.0e-7	7.32E-8
2032-04-30	6.6e-8	-	-	2080-04-27	1.6e-8	2.4e-9	-
2034-04-30	3.6e-7	1.6e-7	-	2081-05-01	3.1e-8	-	-
2036-04-30	1.7e-8	-	-	2090-05-02	1.8e-7	-	-
2038-04-30	1.2e-8	-	-	2090-05-01	8.1e-7	-	7.19E-7
2039-05-01	1.8e-8	-	-	2090-05-01	1.2e-6	-	5.38E-7
2042-05-01	5.4e-9	-	-	2090-05-01	1.3e-6	-	6.31E-7
2043-05-01	1.0e-8	4.6e-9	-	2092-04-28	1.0e-8	-	-
2044-04-30	1.6e-8	-	-	2093-05-01	3.2e-8	-	-
2046-04-30	7.4e-9	-	-	2094-04-28	2.0e-8	-	-
2047-04-27	3.9e-8	3.2e-9	-	2094-04-30	7.0e-9	-	-
2047-04-27	1.2e-7	-	-	2095-04-30	3.9e-8	-	-
2049-04-30	1.2e-8	-	-	2097-04-27	-	-	1.17E-9
2050-04-30	7.9e-9	-	-	2099-04-30	3.1e-8	-	-
2051-05-01	1.7e-7	-	-	2100-04-27	1.1e-8	-	-
2053-04-30	8.7e-9	-	-	2100-04-27	4.6e-6	-	6.27E-7
2056-04-30	9.6e-9	-	-	2100-04-27	1.9e-6	-	2.37E-6
2058-05-01	6.8e-9	-	-	2102-05-01	3.1e-8	-	-
2063-05-02	4.7e-7	2.1e-7	-	2103-04-28	1.2e-8	-	-
2065-04-27	4.7e-8	7.9e-8	-	2105-05-02	2.8e-8	-	- Diservatoire

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